QwkSep 15 Low-Profile Separation System (LPSS)

Design Description

Sierra Nevada Corporation's (SNC) Space Systems QwkSep 15 Low-Profile Separation System (LPSS) provides a low-shock solution to small satellite separation in an extremely low profile. The system is designed for standard ESPA (EELV Secondary Payload Adapter) with a 15-inch satellite interface launch configuration (orthogonal to thrust axis). The interface rings have integrated adjustable kick off springs, pass-through separation connectors and redundant telemetry indication of positive



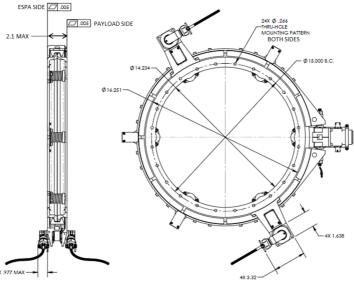
QwkSep 15 Low-Shock Clamp Band. The QwkSep 15 clamp band system provides a small satellite separation solution.

separation. The system is released with a mini, low-shock Clamp Band Opening Device (CBOD). This design configuration has heritage in more than 100 successful flight releases.

The CBOD features redundant circuits driven by a typical pyrotechnic firing pulse. Based on our space-qualified Fast-Acting Shockless Separation Nut (FASSN) technology, the CBOD restrains the band tension bolts with a double helix, flywheel nut. The back drive torque of the high lead, band tension bolts is reacted through the CBOD by the latched flywheel nut. A pyro-compatible pulse releases the flywheel nut, which spins up and ejects the tension bolts. The strain energy in the band is converted to rotational energy in the flywheel nut allowing the two mating halves to separate with extremely low shock.

Feat	Features				
•	Ultra-high reliability payload separation	•	Redundant electrical trigger circuits		
•	>25% stiffer and >40% more load capability than comparable, alternative solutions		Utilizes heritage release technology of CBOD with redundant NASA standard initiator-driven pin puller		
•	Low-shock operation	•	Scalable between 12-inch and 24-inch sizes		
•	Designed for full ESPA payload weight and ESPA dynamic environments		Optional nonpyrotechnic mini-CBOD release mechanism available for extremely low-shock release		
•	Straightforward integration and operation	•	No generated debris		
•	Resettable for multiple ground operations	•	Based on extensive clamp band heritage		

Dimensions



Note: All dimensions above are in inches.



Applications			
Auxiliary payload separation	ESPA-compatible integration and operation		

Heritage Programs					
Nanosat	Orbital Express				
Atlas V*	Delta IV*				
Arianne*	Sea Launch*				
Proton*	• Falcon 9*				

*Note: Larger diameter version (primarily 47-inch and 66-inch systems); have more than 100 combined flight releases on these LVs.

	U.S.	SI		
Mechanical				
Payload capability	400 lbm with 20-inch center of gravity (CG)	181 kg (508 mm CG height)		
	Offset height above ESPA interface			
Quasi-static environment	8.5 g axial and lateral dynamic loading simultaneously			
Random vibration environment	Qualified to NASA General Environmental Verification Specification (GEVS) levels for large (400+lbm) payloads (5.6 grms)			
Stiffness Axial:	2.15E6 lb/in	3.76E4 N/m		
Moment:	9.62E7 in•lb/rad	1.09E7 N•m/rad		
Envelope dimensions	Ø15 BCD x 2.1-inch max. stack height	Ø381 BCD x 53.3 mm		
Mass, full system (not including fasteners, harness)	15 lbm max.	6.8 kg max.		
Mass, flyaway	4.0 lbm max.	1.8 kg max.		
Life (as-delivered)	12 full-load release cycles			
Redundancy	Full electrical			
Source shock	Pyro: 1,000 g from 1 kHz to 2 kHz near actuator			
	Non-pyro option: 100 g max. from 10 Hz to 10 kHz			
Tip-off rate	0.5 °/s max.			
Kick-off rate (separation velocity)	1 ft/s min.	0.3 m/s min.		
Electrical				
Release signal	Pyro: NASA Standard Initiator (NSI)-firing pulse			
	Non-pyro option: 3.5 amps for 50 ms (typical)			
Separation telemetry	Redundant loop-back circuits indicate positive separation			
Release time	50 ms max.			
Thermal				
Operating temperature range	Pyro: -90 °F to +219 °F			
	Non-pyro option: -85 °F to +167 °F			
Reset				
Refurbishment	Replace standard NSI-Pin Puller trigger			
Special tools	SNC band loading tool; SNC spring compression tools			
Time required for reset	~ 2 hours			